CCR Number: 0013 CRITICALITY: URGENT DUE: 03/20/98

DISTRIBUTION SHEET EO-1 LEVEL II CCB

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Randy Harbaugh/Code 511 Dave Folta/Enhanced Formation Fly

John Loiacono/ALI Instrument Engr Don Lencioni/EO-1 Instr. Scientist

NEW MILLENNIUM PROJECT CONFIGURATION CHANGE REQUEST

PROGRAM <u>EO-1</u>	TITLE E	B/L EO-1 TECHN	DLOGY LEVEL II	REQUIREMENTS	DOC.
CCR NO. O R 00	<u>13 N A T C</u>) R	R. C	ARTER/GSFC	
DATE INITIATED O ROS	3/09/98 T C R 'S	C H G . N O			
DUE DATE	SPONSO	OR/CODE	R. CARTER	PHONE	x8421
EFFECTIVITY	CHANGI	E CLASS	TYPE OF CHANGE		
ITEM: ROTS D	oc	1 11	MILESTONE 🔲	INTERFACE \Box	SOFTWARE
S/N	PRELIM	INARY 🗖 🛱	DOCUMENT \Box	POWER \Box	OTHER \Box
	FORMA	└ □□┥	OST u	WEIGHT	
		ENTS OR SOFTWAR			
S/N	EO-1	Level-II Technolog	y Segment Requir	rements	
ITEM:					
S/N					
PROBLEM					
The attached draft v	version of Earth Or	biter-l (EO-1) Tec	nnology Level II R	equirements Docum	nent requires
baselining. This doc	ument contains the	New Millennium	Program (NMP) E	O-1 Technology Le	evel II
Requirements. The	Level II technology	requirements spe	cify and define re	quirements at the i	individual
technology levels. NOTE: Significant ch	nanges with regards	to WIS GIS and	FODB are curren	tly being considered	d This CCR
does not address the					
goal here is to basel	line the requirement	s document and a	dd changes via the	CCR process at a	a later date.
PROPOSED SOLUTION					
Approve the attached					
Configuration Contro					
Requirements docum (CCRs) and Prelimin			•	•	•
maintained by the E				p. 6 . a	
BOARD ACTION	APPROVAL LEVEL	CRITICALITY LEVE	L PROCUREMEN	NT CHANGE ORDER (CLASSIFICATION
APPROVE \Box	REQUIRED LEVEL I HQS	EMERGENCY	ROUTINE	URGENT	EMERGENCY
APPROVE WITH CHANGE	LEVEL II GSFC	URGENT 2	OPTION 1	OPTION 1 \Box	
DISAPPROVE	LEVEL III	ROUTINE	OPTION 2	OPTION 2	
withdraw					
COMMENTS		l			
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	CHAIRPERSON	WAT		DATE 22.	Jun 98
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New Millennium Program/Earth Orbiter-1 Technology Level II Requirements

1.0 Introduction

This document contains the New Millennium Program (NMP)/Earth Orbiter-1 (EO-1) Technology Level II Requirements. The Level II EO-1 technology requirements specify and define requirements at the individual technology levels. These requirements address technology level functional and performance specifications that define success criteria for technology demonstrations. Also included in this document are the interface requirements to the spacecraft and ground segment.

The EO-1 mission requirement definition is accomplished in three levels. The Level I requirements define the EO-1 mission objectives and products. The Level II requirements identify and allocate appropriate requirements to mission segments (Technology, Spacecraft, Ground Segment). The Level II requirements are top level requirements for each mission segments. The Level III requirements are the lowest level requirements for the mission. The Level III requirements are directly implemented at the hardware and software levels. The Level III requirements are traced to Level II and then to Level I, respectively. All Level III requirements have either parent requirements in Level II and/or Level I, or have justification for its orphan status.

2.0 Requirement Organization

Requirements are organized and identified by Requirement ID, Requirement Type, Requirement Title, and Requirement Statement. The Requirement ID is a numbering system where each requirement is assigned a unique number. This number is used in tracing a requirement from parent to child and vice versa. The Requirement Type is an indicator for a type of requirement. The detailed description of the Requirement Type is provided in the section 3.0. The Requirement Title is a title for a requirement. The Requirement Statement provides the required action or activity. There is only one required action or activity per requirement statement.

3.0 Requirement Type Definition

H (**Hierarchical**) **Requirement**: A requirement which is not directly verifiable, but provide structure to a set of requirements. A Hierarchical requirement must be verified "through validation." This means that the child requirements must be validated to define the success of the parent, and the child requirements must be verifiable. (Note: Child requirements may themselves be hierarchical.)

FC (**Functional Category**): A hierarchical requirement, which is the parent of a set of child requirements. Generally the functional category may be viewed as a container of a set of requirements which are "alike" in some manner such as; a. Similar Functionality

- b. Same functionality applied to different elements
- c. Are verified as a group.

F (Functional) Requirement: A functional requirement is a child requirement to a functional category. A functional requirement is the parent requirement to performance requirements. Functional requirements specify functions of the system, subsystem, instrument, or component. Functional requirements must be verifiable by test, analysis, or inspection.

P (**Performance**) **Requirement**: A performance requirement is a child requirement to a functional requirement. The performance requirements are directly verifiable, and each performance levels are verified. The performance requirements specify discrete performance levels of the system, subsystem, instrument, and/or component. The verification methods for performance requirements are test, analysis, and/or inspection.

4.0 Requirement Verification

The requirements shall be verified using methods accepted by the EO-1 Mission management. The acceptable verification methods include testing, analyses, and/or inspection. Positive verification for each requirement shall be provided. A requirement shall be verified either directly or indirectly. An example of an indirect verification would be such that Level I requirement is traced to Level II and then to Level III, where a direct verification of the Level III requirement is accomplished. In this case, the Level I and II requirements are verified indirectly and the Level III requirements are verified directly. This is an acceptable verification approach.

5.0 Requirements Verification Matrix

The requirement verification matrix shall be developed. The matrix shall identify requirement, verification method, verification acceptable criteria, verification results, and the date of verification.

6.0 EO-1 System Validation

The requirement verification matrix, along with requirements tracing to either parent or child shall be the basis for the EO-1 system validation.

Requirement ID		Requirement Title	Requirement Statement
01	nt Type H	NMP/EO-1 Technology: Level	This document defines Level II requirements for the New Millennium Program (NMP) EO-1 Technologies.
01.01	Н	General	All technologies shall adhere to the requirements in this section as applicable.
01.01.00.01	F	Mission Life	Shall be designed to support mission life of 1 year nominally and 18 months for expendables
01.01.00.02	F	Technology Category "II"	Technologies that are in Category "II" shall have alternates ready at the time of EO-1 observatory integration and test.
01.01.00.03	F	Environment	All hardware and software shall adhere to EO-1 Environmental Requirements.
01.01.00.04	F	Mission Assurance	All hardware and software shall adhere to EO-1 Mission Assurance Requriements.
01.01.00.05	F	Verification	Shall provide for adequate visibility to accommodate effective subsystem and system functional and performance verification at all stages of development.
01.02	Н	Multispectral	Shall demonstrate superior multispectral imaging capability.
01102		Imaging Capability	enan demenerate capener manapeerar magnig capability.
01.02.01	FC	Images	Shall ensure continuity of the Landsat 7 data set by collecting landsat- type MS/PAN images.
01.02.01.01	F	Image Type	Shall gather Landsat-type multispectral terrain images capturing seasonal variations encompassing one entire growing season (March thru October) in Northern Hemisphere.
01.02.01.02	F	Landsat Type	Shall gather Landsat type spectral bands from 0.4 um to 2.5 um.
01.02.01.03	F	Prime Spectral	Shall gather prime bands (TBS um).
01.02.01.04	F	Telescope	The telescope optomechanical performance shall provide unobstructed optics that allow substantially small focal plane diode.

Requirement ID	Requireme nt Type	Requirement Title	Requirement Statement
01.02.01.05	F	Focal Plane	MS/PAN focal plane module shall demonstrate the mosaic of 4 Sensor Chip Assembly (SCA), each SCA having a spectrum band and pan band of 0.48 to 0.68 um.
01.02.02	FC	Calibration	Shall evaluate calibration capabilities.
01.02.02.01	F	Parameters	Shall support calibration goal of 5% radiometric for future missions.
01.02.02.02	F	Design	Shall have variable apertures, diffusers, and internal sources to support the calibration goals.
01.02.02.03	F	Calibration	Sun, moon, and deepspace shall be viewed during calibration.
01.02.03	FC	Paired Scene	Shall support 200 paired scene comparisons with Landsat 7.
01.02.03.01	F	On-orbit operation	The ALI focal plane electronics shall operate for 10 minutes maximum per daylight period of each orbit, up to four times per 24 hour period.
01.02.03.02	F	Ground Formation	Shall provide ground-based formation flying of sufficient precision.
01.02.04	FC	MS/PAN Focal Plane Thermal	Shall demonstrate non-cryogenic capability of the near and short wavelength infrared detectors.
01.02.04.01	F	HgCdTe Detectors	Demonstrate operation of HgCdTE detectors at the nominal 220 degree K thermal environment.
01.02.04.02	F	Thermal Stability	Provide the necessary thermal stability to achieve stable focal plane detector operation.
01.03	Н	Wide Field, High Resolution,	Shall provide the basis for a Landsat equivalent multisepctral swath width and resolution.
01.03.00.01	F	Optical Design	Optical design shall yield Landsat cross-track field with an in-track field to accommodate the detectors.
01.03.00.02	F	Optical Presciption	Optical prescription shall yield sufficent performance to support 10 m pan band.

Requirement ID	Requireme nt Type	Requirement Title	Requirement Statement
01.03.00.03	F	Push Broom Mode	Shall operate in pushbroom mode covering entire Landsat-7 swath of 185 Km.
01.03.00.04	F	Cross Track Coverage	MS/PAN focal plane sensor chip assemblies shall permit Landsat cross-track coverage with no gaps.
01.04	Н	Silicon Carbide Optics	Shall Provide the basis for reflective optical systems that are light and stable over a wide range of operating temperatures.
01.04.00.01	F	Large Aperture	Telescope shall demonstrate large aperture SiC optics.
01.05	Н	Wedge Imaging Spectrometer	Shall provide the basis for the future hyperspectral imaging system.
01.05.01	FC	Images	WIS images shall be collected.
01.05.01.01	F	Focal Plane	WIS and GIS shall view the same ground track.
01.05.01.02	F	Ground Sample Distance	The ground sample distance shall be same as Landsat ground sample distance.
01.05.02	FC	WIS Data	WIS data shall be evaluated.
01.05.02.01	F	Data Synthesis	Shall synthesize Landsat data, 0.4 um to 2.5 um, with the WIS data.
01.06	Н	Grating Imaging Spectrometer	Shall be used to critically evaluate the time dependent spectral sampling of the WIS.
01.06.01	FC	Images	GIS images shall be collected.
01.06.01.01	F	Focal Plane Placement	GIS shall view the same ground track as WIS.
01.06.01.02	F	FPA	GIS FPA design and data format shall allow for WIS to GIS comparison.

Requirement	•	Requirement Title	Requirement Statement
ID	nt Type		
01.07	Н	Atmospheric Corrector (AC)	The Atmospheric Corrector (AC) shall enhance land imaging.
01.07.00.01	F	Image Collection	Shall collect TBD number of images to perform spectral and spatial characterization.
01.07.00.02	F	Calibration	Solar, lunar, and deepspace scans shall be used for radiometric calibration.
01.07.00.03	F	Relative Pointing	Shall be capable of determining the relative pointing of the ALI and the AC as defined in ICD.
01.07.00.04	F	Operation	AC images shall be taken while MS/PAN, WIS & GIS, and Landsat images are being collected to provide ene-to-end system validation.
01.08	Н	Fiber Optics Data	FODB shall serve as a data transmission media, electronics and
		Bus (FODB)	protocol for science data between and ALI, the AC, and the WARP.
01.08.00.01	F	Data Transmission	FODB shall provide high quality science data transmission through the FODB interface.
01.08.00.01.0	Р	Data Rate	FODB shall be capable of transferring at rates up to 1.0 Gbps.
01.08.00.01.0	Р	Error Rate	FODB error rate shall be 1.0E-09 or less.
01.09	Н	X-Band Phased	The XPAA shall demonstrate light weight antenna that is electronically
		Array Antenna	steerable.
01.09.01	FC	Science Data Transmission and	Shall be capable of transmitting science data and link error shall be established.
01.09.01.01	F	Science Data Transmission	Shall provide a 105 Mbps QPSK data link for science data return to earth.
01.09.01.02	F	Bit Error Rate (BER)	Measurement and tabulation of basic BER shall be taken during science downlink.
01.09.01.03	F	Error Burst Length	Error burst length data shall be taken during science downlink.
01.09.02	FC	Antenna Pattern	The antenna pattern scan shall be performed periodically.

Requirement ID	Requireme nt Type	Requirement Title	Requirement Statement
01.09.02.01	F	Antenna Gain Measurements	Main lobe antenna gain measurements shall be taken on the ground and in space.
01.09.02.02	F	Controller	Shall demonstrate the reliability of the software and controller of the array.
01.10	Н	Enhanced Formation Flying	The EFF shall provide the autonomous capability of flying over the same ground track of another S/C at a fixed separation in time.
01.10.01	FC	Autonomy	The EFF shall provide on-board autonomous relative navigation and formation flying control for the EO-1.
01.10.01.01	F	AutoCon Flight Control System	The AutoCon flight control system shall provide autonomous formation flying control.
01.10.02	FC	Ground Track	EO-1 shall fly over the same ground track (+/- 3Km) as Landsat-7.
01.10.02.01	F	Separation	Shall maintain a 1-minute in-track separation between EO-1 and Landsat-7.
01.11	Н	Light Weight Solar Array (LWSA)	The LWSA shall provide the basis for future light weight solar panels.
01.11.00.01	F	Deployment	Shall demonstrate a controlled deployment using Shape Memory Alloy hinges.
01.11.00.02	F	Efficiency	Shall demonstrate specific energies greater than 100 W/Kg.
01.11.00.03	F	Dynamic Performance	Dynamic performance shall be evaluated using accelerometer during spacecraft yaw manuver.
01.12	Н	Carbon-Carbon Radiator (CCR)	The CCR validation shall provide the basis for radiators that are considerably lighter and have greater thermal conductivity than aluminum.
01.12.00.01	F	Thermal	Thermal conductivity shall be evaluated.

Requirement	Requireme	Requirement Title	Requirement Statement
ID	nt Type		
01.12.00.02	F	Mechanical	CCR mechanical properties shall meet the structural requirements of
		Properties	the S/C design.
01.12.00.03	F	Contamination	The CCR shall be encapsulated to preclude contaminating ALI.
01.12.00.04	F	Thermal Dissipation	Thermal dissipation shall be 60 W per ICD.
01.13	Н	Pulsed Plasma	The PPT validation shall provide the basis for a low mass, low cost,
		Thruster (PPT)	highly reliable, and safe propulsion system.
01.13.00.01	F	Control Capability	The PPT shall demonstrate capability to replace all science mode functions of pitch wheel.
01.13.00.02	F	Demonstration	The PPT shall be demonstrated and validated after ALI has been validated.
01.13.00.03	F	Plume	Shall confirm that the PPT plume is benign to the optical surfaces of the ALI.

Date: Tue, 10 Mar 1998 08:42:49 -0500 (Eastern Standard Time)

From: Administrator@hst-nic.hst.nasa.gov

Reply-to: (Nicholas Speciale)

Subject: CCR:0013 - DUE: 03/20/98 URGEN Level-2 Nicholas Special WWW-COMMENTS

USER: (Nicholas Speciale) sent the following comments on:

Date: 03/10/1998 CCR Number: 0013 Sponsor: R. CARTER Due Date: 03/20/98

CCR Title: B/L EO-1 TECHNOLOGY LEVEL II REQUIREMENTS DOC.

Remote host: 128.183.212.178 Email Address:

APPROVAL STATUS: APPROVED

Note

COMMENTS:

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01.02.01	FC	Images	Shall ensure continuity of the Landsat 7 data set by collecting landsat- type MS/PAN images.
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01.02.01.04	F	Telescope	The telescope optomechanical performance shall provide unobstructed optics that allow substantially small focal plane diode.

Rith, we need a couple reservence documents to apply to all. (Mechanical, electrical, Anduct Assurance, and such) Please Add These.

List The Document

T'm Not sure how Thin can realistically be verified

These belong in The GND section Requirements

Fruide The Value

Requirement	Requireme	Requirement Title I	Requirement Statement
ID	nt Type		
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01.02.03	FC	Paired Scene	Shall support 200 paired scene comparisons with Landsat 7.
01.02.03.01	F	On-orbit operation	The ALI focal plane electronics shall operate for 10 minutes maximum per daylight period of each orbit, up to four times per 24 hour period.
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01.02.04.02	F	Thermal Stability	Provide the necessary thermal stability to achieve stable focal plane detector operation.
31.03	Н	Wide Field, High	Shall provide the basis for a Landsat equivalent multisepctral swath
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31.03.00.01	F	Optical 'Design'	Optical design shall yield Landsat cross-track field with an in-track field to accommodate the detectors.
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I'm not sure
what This is
ment to
say, please
see me re: This.
Actually John can
rephrase it

in behind L7 and ±3km cross track)

Requirement	Requireme	Requirement Title I	Requirement Statement	
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01.07.00.01	F	Image Collection	Shall collect TBD number of images to perform spectral and spatial characterization. STET (Demis to Proce TBD)	coverage and spectral
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01.07.00.03	F	Relative Pointing	Shall be capable of determining the relative pointing of the ALI and the AC as defined in ICD.) I don't understand The requirement
01.07.00.04	F	Operation	AC images shall be taken while MS/PAN, WIS-& GIS, and Landsat- images are being collected to provide en to-end system validation.\	-
				L7 takes >250 Scenes per day
01.08	Н	Fiber Optics Data Bus (FODB)	FODB shall serve as a data transmission media, electronics and protocol for science data between and ALI, the AC, and the WARP.	so we can't
01.08.00.01	F	Data Transmission	FODB shall provide high quality science data transmission through the FODB interface.	actually da This
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01.12.00.01	F	Thermal	Thermal conductivity shall be evaluated.

can we do This? Check w/ Ken Perko.

		Requirement Title	Requirement Statement	
ID	nt Type			
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		Properties	the S/C design.	
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01.12.00.04	F	Thermal Dissipation	Thermal dissipation shall be 60 W per ICD.	- List The ICD Name
	1			
01.13	Н	Pulsed Plasma Thruster (PPT)	The PPT validation shall provide the basis for a low mass, low cost, highly reliable, and safe propulsion system.	
01.13.00.01	F	Control Capability	The PPT shall demonstrate capability to replace all science mode functions of pitch wheel.	
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01.13.00.03	F	Plume	Shall confirm that the PPT plume is benign to the optical surfaces of the ALI.	

Page 1 of 1

CCR SPONSOR RECOMMENDATION FORM

CCR NUMBER: 0013

CCR TITLE: B/L EO-1 TECHNOLOGY LEVEL II REQUIREMENTS

CCR SPONSOR: Ruth Carter/GSFC

SUMMARY OF COMMENTS RECEIVED: (list Level 4 CCB and internal

reviewers who had comments and address those comments)

Pete Spidaliere: See hard copy red lines.

Sponsor Recommendation: Incorporate all of the recommended changes to document which includes the attached red lines.

SPONSOR/ORGANIZATION: Ruth Carter/GSFC

DATE: 5/14/98